

Stirling Converter (Engine) Power Generation System: Control Electronics

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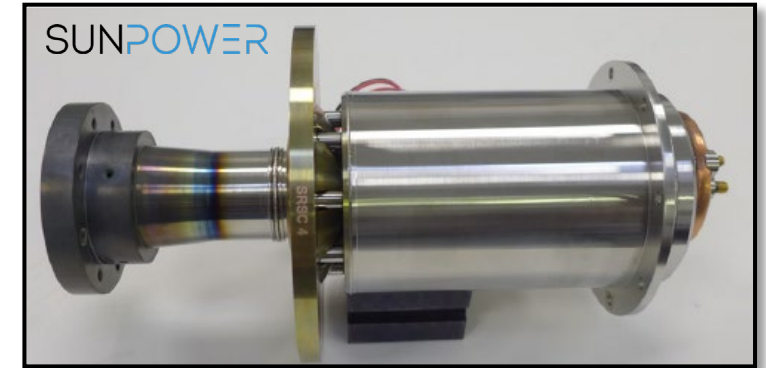
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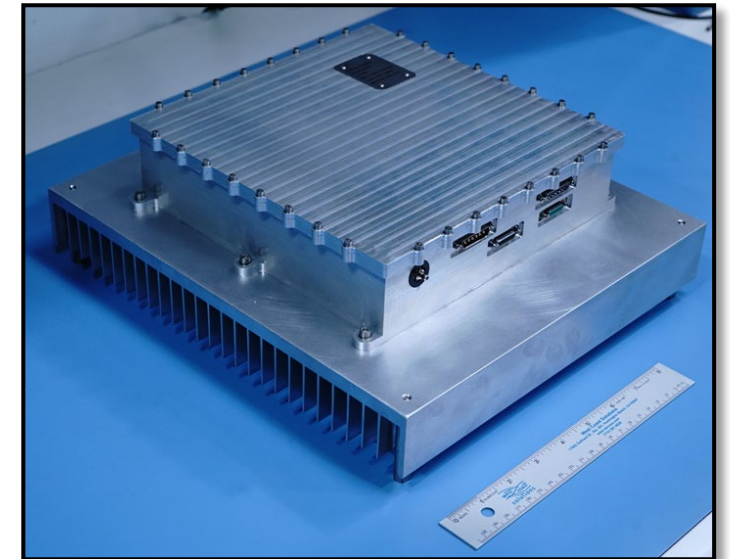
Background and Motivation

- Radioisotope Power Systems (RPS) consist of a heat-generating isotope power source in combination with a thermal / electrical power conversion system
 - Legacy RPS systems utilized thermoelectric conversion with efficiency **~6%**
 - Stirling power conversion efficiency currently approaches **30%** → *less isotope required for a given power level*
- High-efficiency, Stirling RPS are garnering interest for multiple applications, including Deep space probes, rovers and landers, and lunar Survive the Night applications
- Stirling Convertors themselves are approaching maturity, though they must be paired with a suitable space-compatible, radiation hardened controller system
 - **Controller / Convertor interactions are nontrivial, and demonstration of a path-to-flight controller with a Stirling Convertor provides substantial risk reduction for a variety of current and future programs**
- AFRL and Intuitive Machines are planning to fly a Sunpower Convertor and WCS Controller on the Space Operations via Nuclear Emerging Technologies (SONET) Program
- Initial demonstration flight has been dubbed Stirling Technology spAce Research experiment (START)

Intuitive Machines and AFRL sponsored a critical risk reduction test to prove the viability of the combined path-to-flight SRSC / SCSC power conversion system



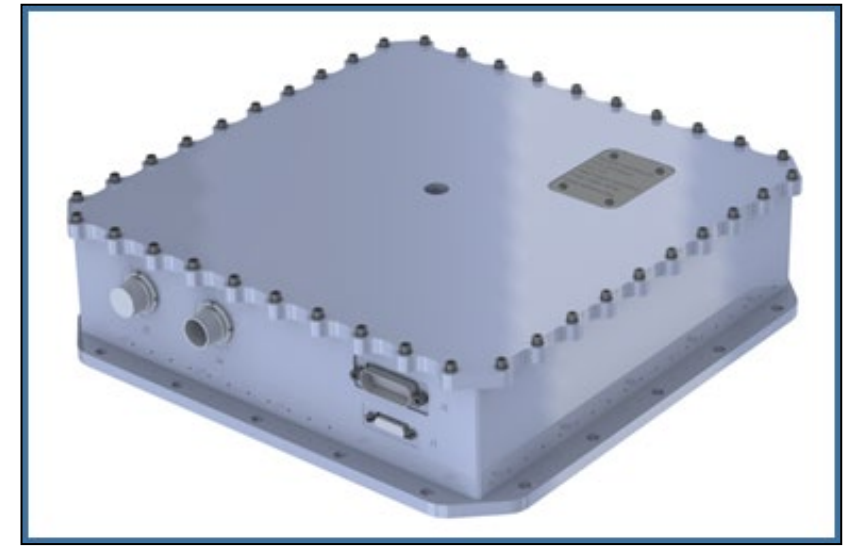
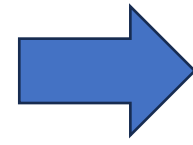
Sunpower Robust Stirling Convertor (SRSC)



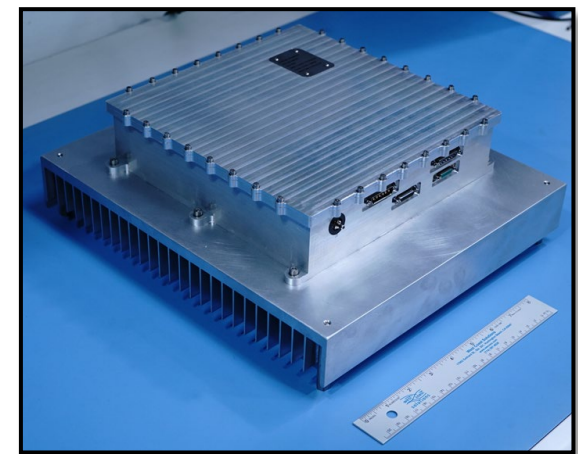
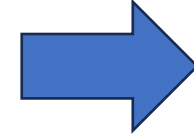
Stirling Convertor Space Controller (SCSC): High Fidelity Brassboard Unit

Stirling Convertor Space Controller(SCSC)

- WCS has developed a spaceflight, radiation hardened Stirling Convertor Space Controller (SCSC) through the NASA SBIR Program (Phase II and Phase III)
- Engineering Model Qualification Unit is in assembly with Qual Testing scheduled for June 2026
- Flight Model is at CDR level – final assembly pending successful EM Qual
- High fidelity Brassboard (BB) SCSC from the Phase I SBIR used for the Risk Reduction testing due to availability
- The BB SCSC contains all of the necessary functionality for the planned testing, and was designed with a straightforward path-to-flight
- BB SCSC circuit board integrated with a convective heatsink for the planned benchtop testing



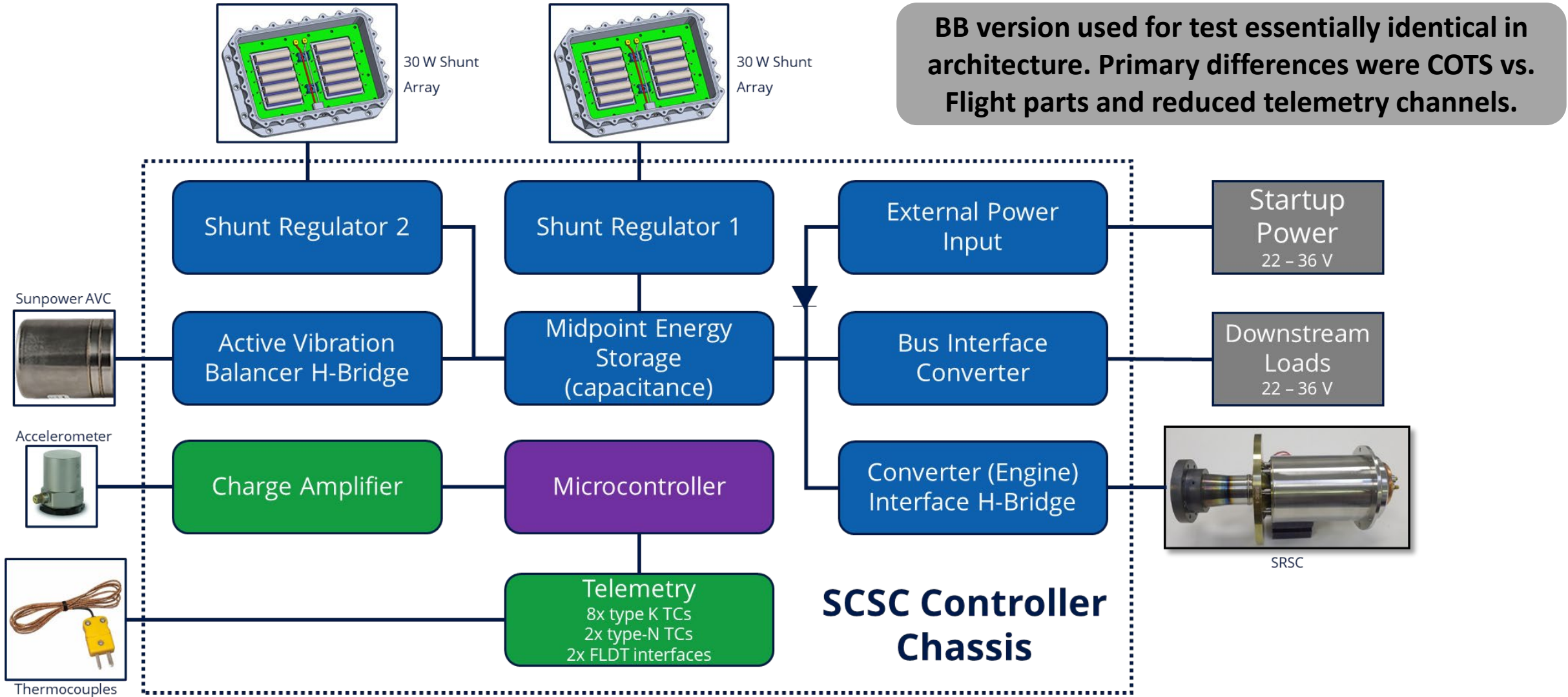
EM/FM SCSC. 10" x 10" x 2.8"; ~ 4kg



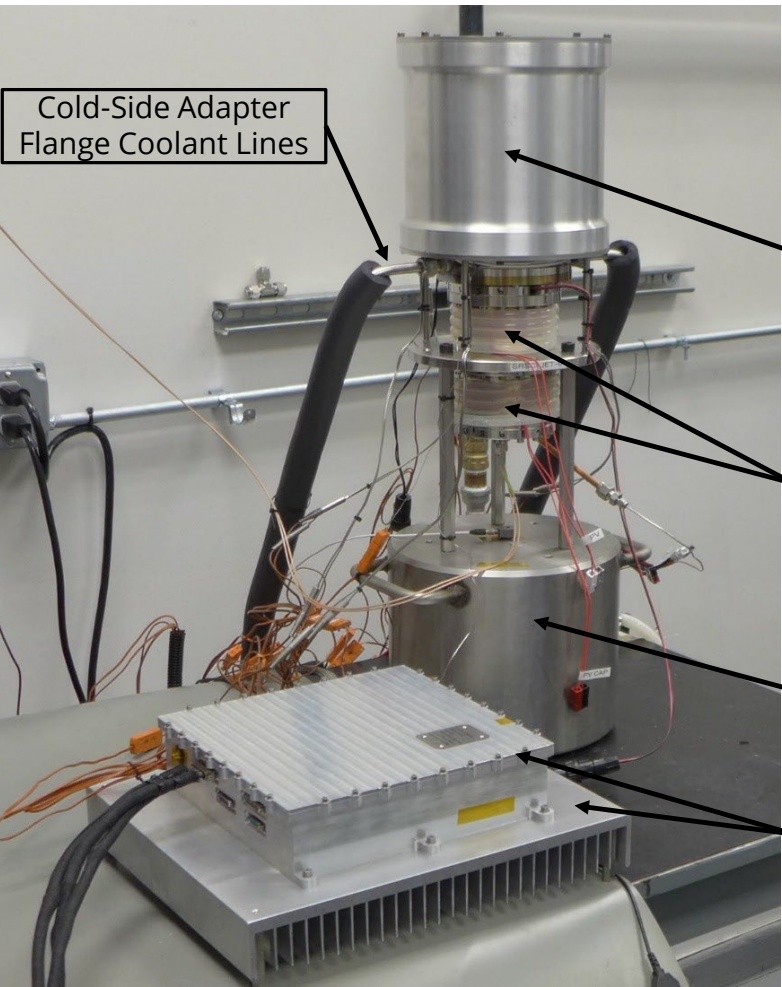
BB SCSC used for risk reduction testing in October 2025.

The BB SCSC is highly representative of the Flight Model SCSC

Flight SCSC Functional Block Diagram



Sunpower Test Facility and Setup



Cold-Side Adapter
Flange Coolant Lines

Heater /
insulation for
SRSC Hot-End

SRSC Pressure
Vessel with coolant
/ heating lines

SRSC Mounting
Base

BB SCSC with
test chassis
and STE
Heatsink

Test Results: Steady-State Mapping

300 °C
0 °C reject

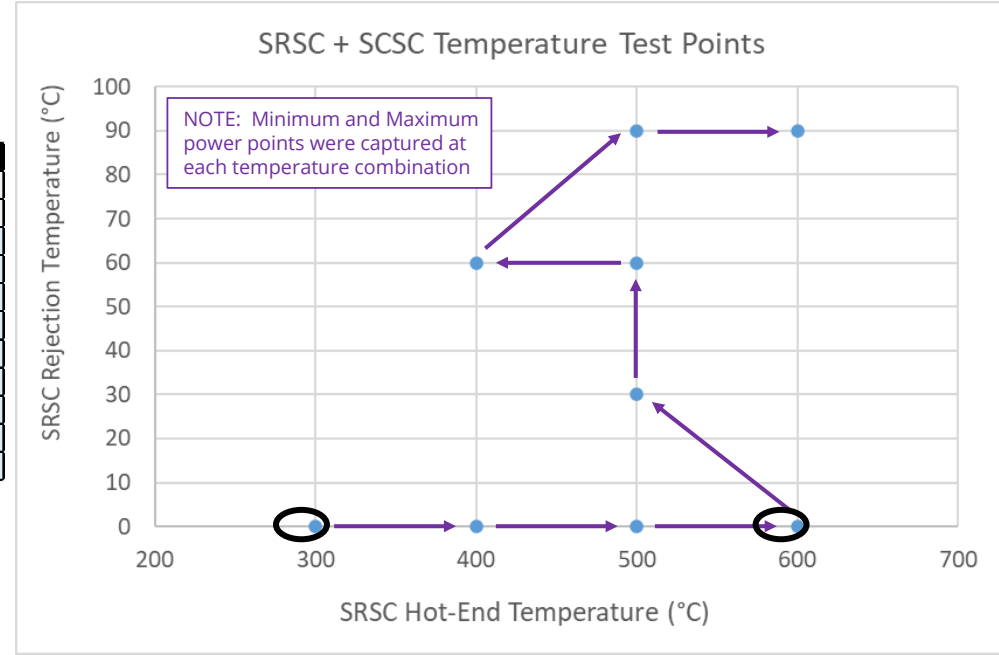
600 °C
5 °C reject

SRSC / SCSC Steady-State Testing All points recorded on 10/29/2025

SRSC Hot-End Temp [C]	299.2	301.1	399.9	399.9	498.9	501.1	599.6	598	500.3	500.9	500.4	498.2	498.9	402.2	401	498.9	497.9	600.8	599.6
SRSC Rejection (CSAF) Temp [C]	0.2	2.1	2.6	0.8	1.9	4.4	5.1	2.4	29.9	30	30.2	60.3	59.7	59.8	60.5	90.9	89.3	90.8	89.7
CSAF Temp Control Cart Setting [C]	-10	-22	-22	-10	-10	-23	-24	-11	19.5	12.3	6	36.5	50.5	51	38	68	81	81.5	65.5
SRSC Alternator Stroke [normalized]	1.06	2.135	2.145	1.06	1.08	2.22	2.245	1.08	1.08	1.7	2.135	2.185	1.07	1.09	2.18	2.19	1.045	1.085	2.17
SRSC Displacer Stroke [normalized]	0.84	1.5	1.57	0.875	0.92	1.67	1.73	0.935	0.94	1.36	1.645	1.7	0.95	0.935	1.64	1.725	0.945	1.005	1.77
SCSC Chassis Temp [C]	23.5	24.4	24.9	24.5	24.1	24.6	25.2	24.9	24.2	24.5	24.9	25.2	24.9	24.7	24.7	25	25	24.8	25.2
SCSC PCBA Temp [C]	24.2	25.9	26.7	25.6	25.3	26.7	27.7	26.4	25.6	26.2	27	27	26.2	25.6	25.9	26.7	26.2	26.2	27.6
SCSC Terminal AC Power [W]	8.83	23.15	38.53	12.44	15.4	52.74	63.46	16.93	15.1	33.82	49.47	48.81	14.12	11.92	35.75	44.5	12.39	15.17	55.2
SCSC DC Bus Power Output [W]	2.65	15.38	30.08	5.95	8.5	43.25	52.94	9.67	8.26	25.73	40.56	40.25	7.38	5.6	28.36	36.3	5.75	7.92	45.3
SCSC DC Bus Voltage Output [V]	4.77	11.44	15.93	7.07	8.46	19.04	21.06	8.97	8.29	14.66	18.42	18.35	7.84	6.83	15.39	17.42	6.92	8.12	19.5
SCSC DC Bus Current Output [A]	0.56	1.34	1.89	0.84	1.01	2.27	2.51	1.08	0.99	1.75	2.2	2.19	0.94	0.82	1.84	2.08	0.83	0.97	2.33

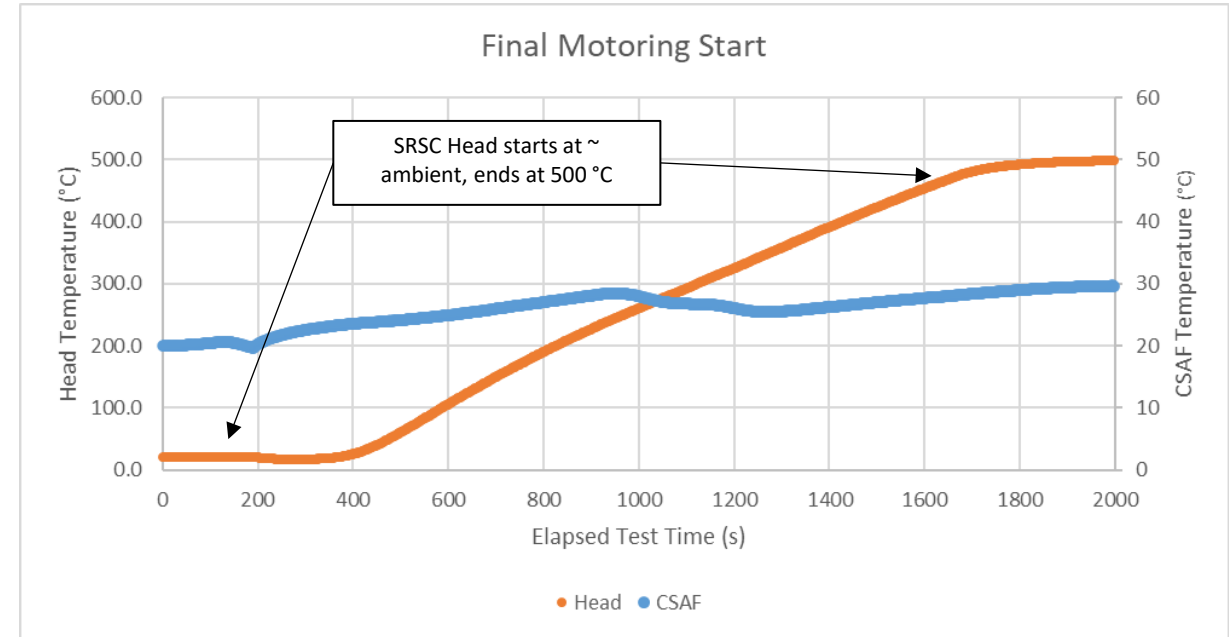
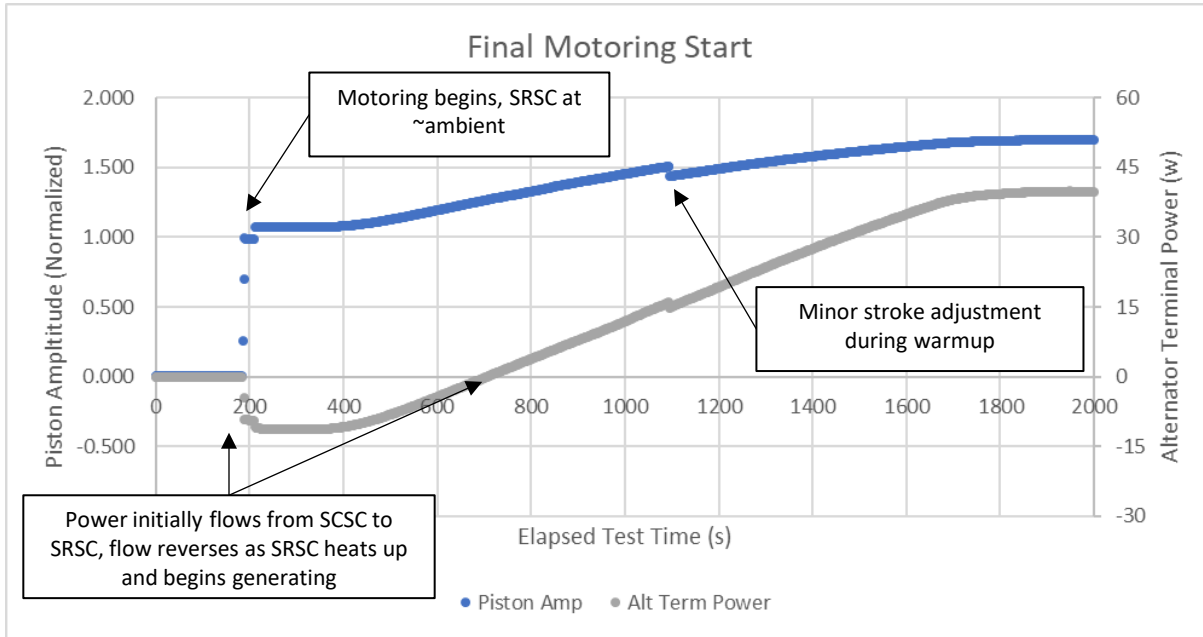
Lowest Power:
8.8 Wac

Highest Power:
63.5 Wac



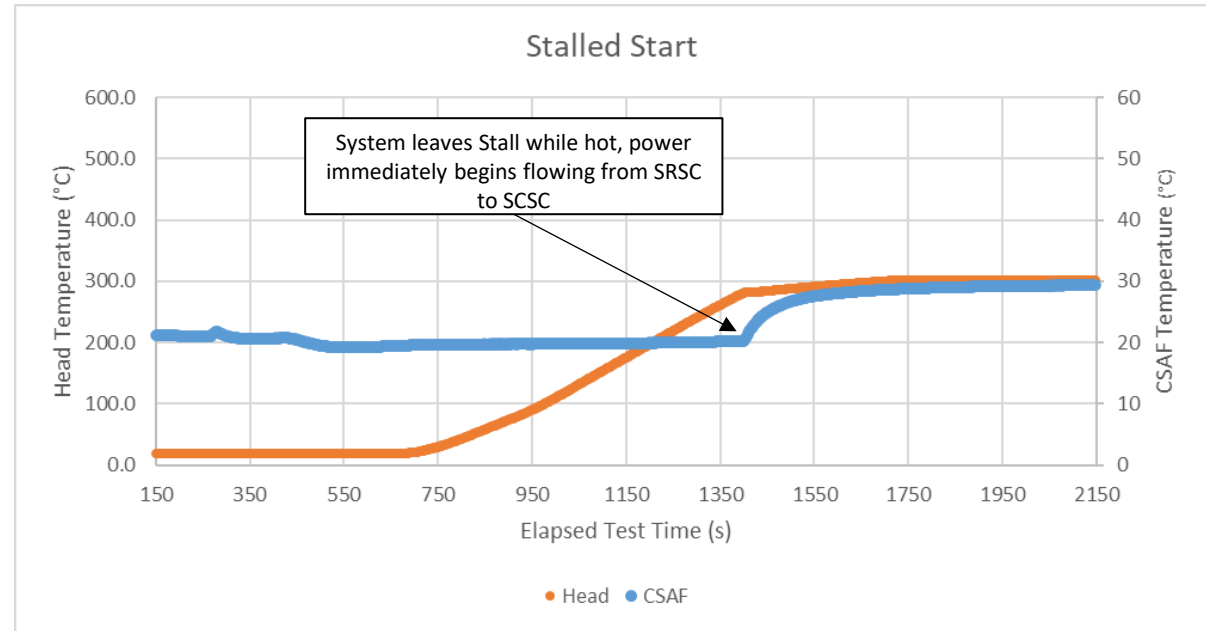
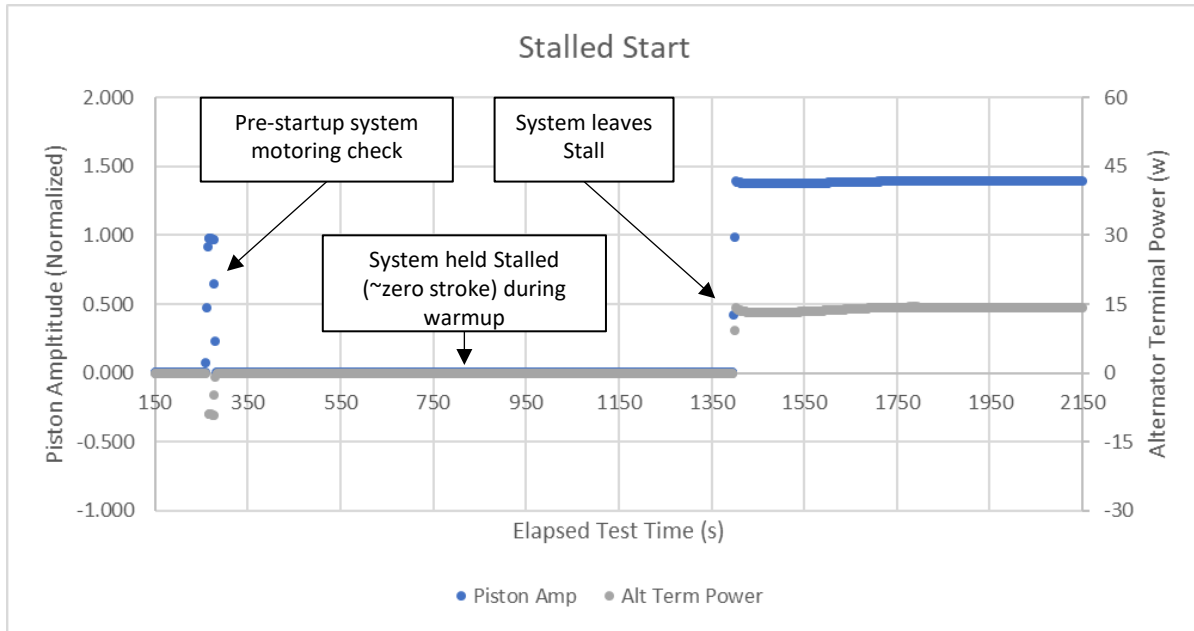
Steady-State testing was performed across a wide range of source (hot end) and rejection temperatures. SRSC AC power levels (input to the SCSC) range from ~9 W to ~63 W.

Test Results: Motor Startups



Motoring-type startup procedures were utilized twice in the testing, no issues in either case

Test Results: Stalled Startup

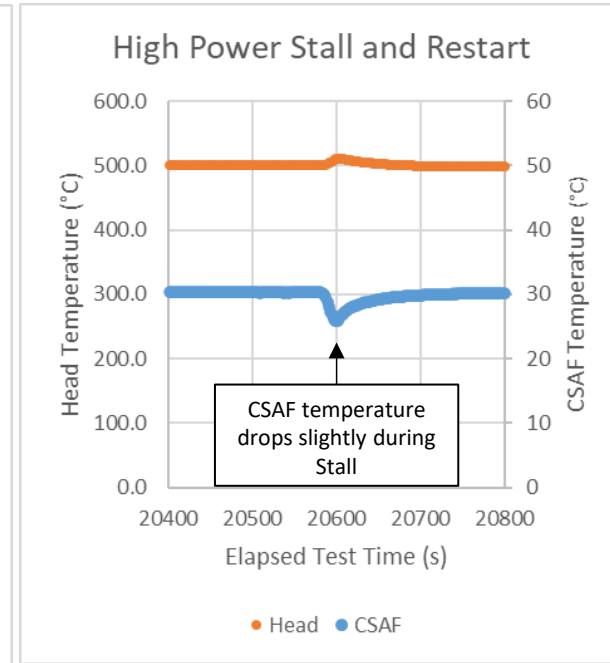
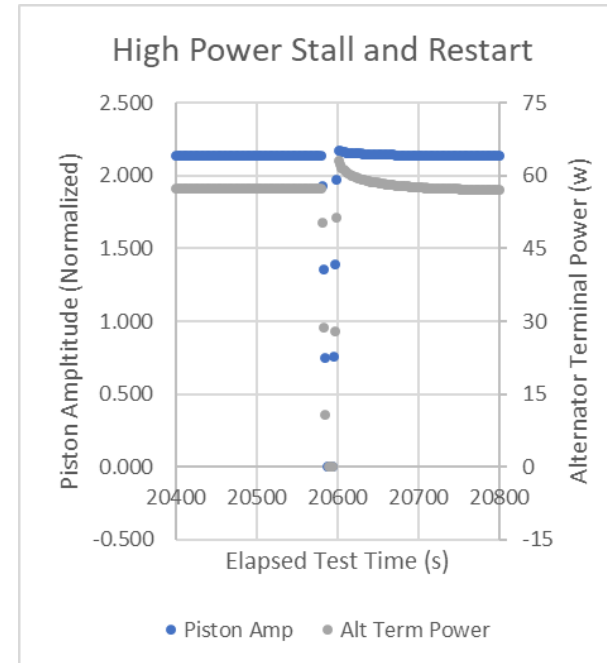
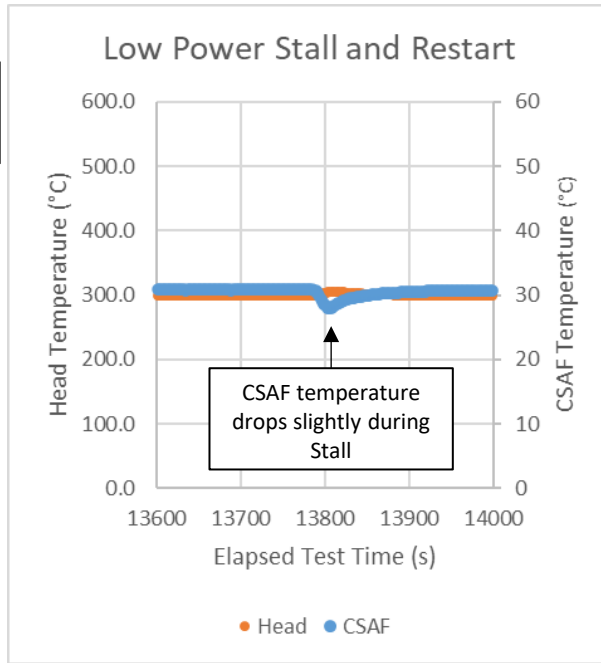
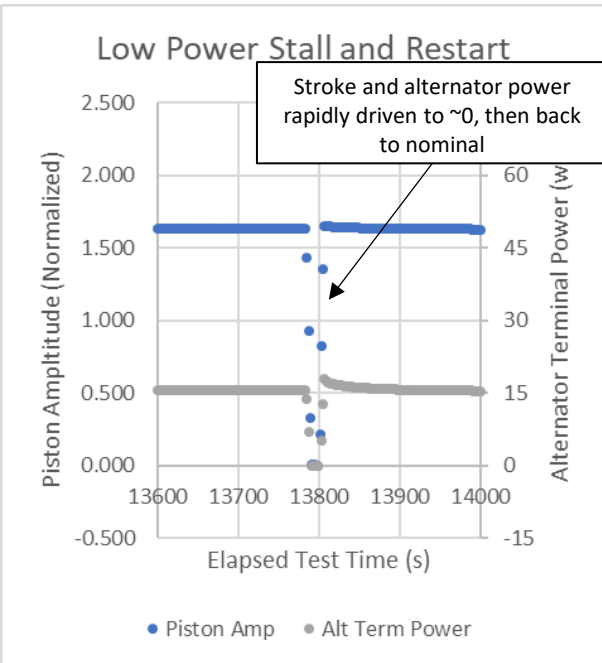


System was held in Stalled state during warmup, piston stroke ramped up at Head temperature of ~300 °C with power immediately being delivered by SRSC to SCSC

Test Results: Stall and Restart Demonstration

Low-Power demonstration at 300 °C and 15 Wac

High-Power demonstration at 500 °C and > 50 Wac



Stall / Restart procedure demonstrates SCSC ability to induce SRSC stall in a hot system, and subsequent controlled restart; stable control demonstrated

Conclusion

- This test marked the first time a path-to-flight controller has been used to drive an actual Stirling Convertor (to our knowledge)
- All test goals met with significant demonstrations achieved
 - Motoring and Stalled startup procedures
 - Hot stall-and-restart
 - System shutdown
 - Accumulated steady-state operation time
- SRSC / SCSC highly scalable to meet higher power demands
 - Sunpower had demonstrated up to 6 kW Convertors
 - WCS under contract to develop an 8-channel SCSC (>500 W)
 - Higher power single- and multi-channel Controllers on the WCS Road Map

Stirling Convertor / Convertor Controller Power Chain Successfully Demonstrated – Proving Viability for a Wide Range of Applications

