

Development of a Synchronized Deployment Solar Array Wing for SmallSats

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WHO WE ARE



DHV Technology is a Spain-based international company that **designs** and manufactures solar panels and other power subsystems for space applications.

DHV Technology supplies solar panels and fully customized solutions for the main international companies in the space sector.

Our facilities, with a total of 3700 m², consist of:

- + 1200 m² clean room
- + 1000 m² offices
- + 1500 m² warehouse and others



HERITAGE

350+

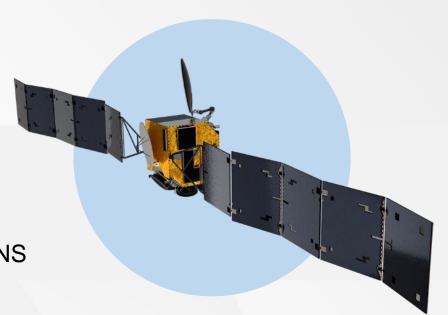
PROJECTS
CARRIED OUT

3000+

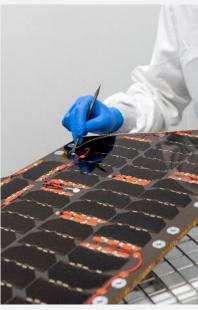
ACCUMULATED DAYS IN ORBIT

330+

SATELLITES FLYING WITH OUR SOLUTIONS











Pre-Development Context

- Larger solar array wings demand for the SmallSats market
- Hinge motorization adjustment required during integration
- Electrical wire bundles become stiffer
- Confirmation of full deployment in flight by signals
- Controlled and synchronized deployment required



SYNCHRONIZED DEPLOYABLE SOLAR ARRAY WING

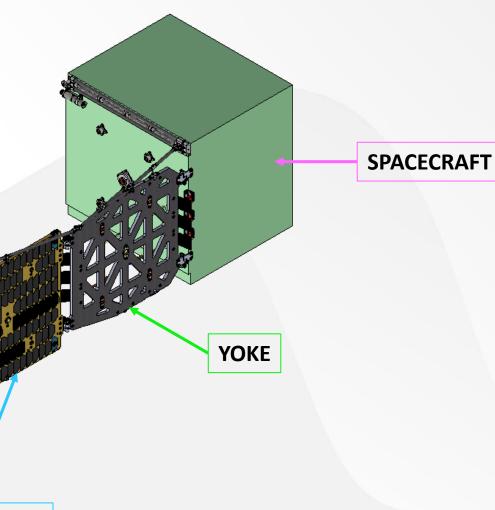
Features:

Yoke + 3 Solar Panels

Clockspring driven hinges

Flexible PCB interconnections

Synchronized and Damped Deployment



(SAW)

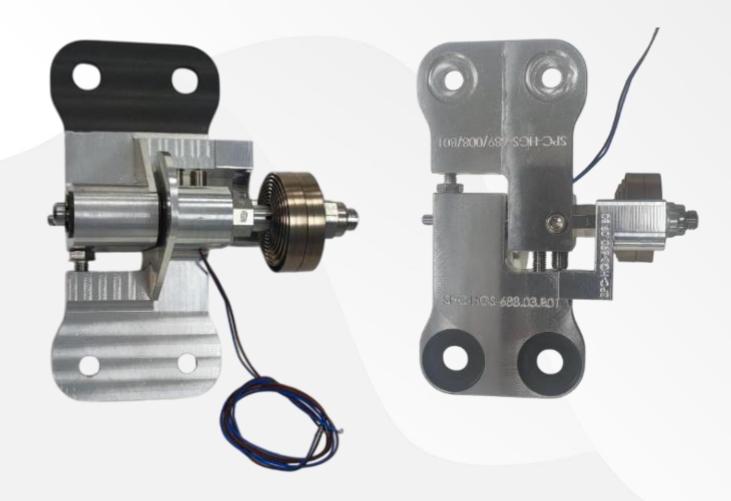
SOLAR PANELS



SAW CLOCKSPRING DRIVEN HINGE

Features:

- Clocksprings → Adjustable torque
- Latching pin
- Adjustable stoppers → Reduce backlash
- Electrical insulation
- Deployment sensor

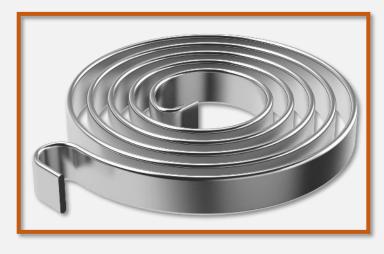


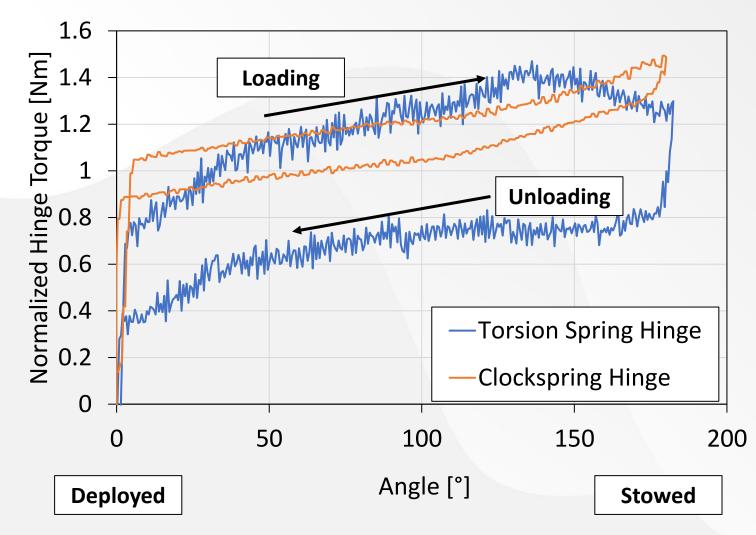


SAW CLOCKSPRING DRIVEN HINGE









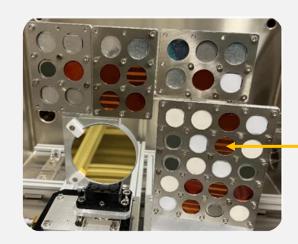


SAW FLEXPRINT

Features:

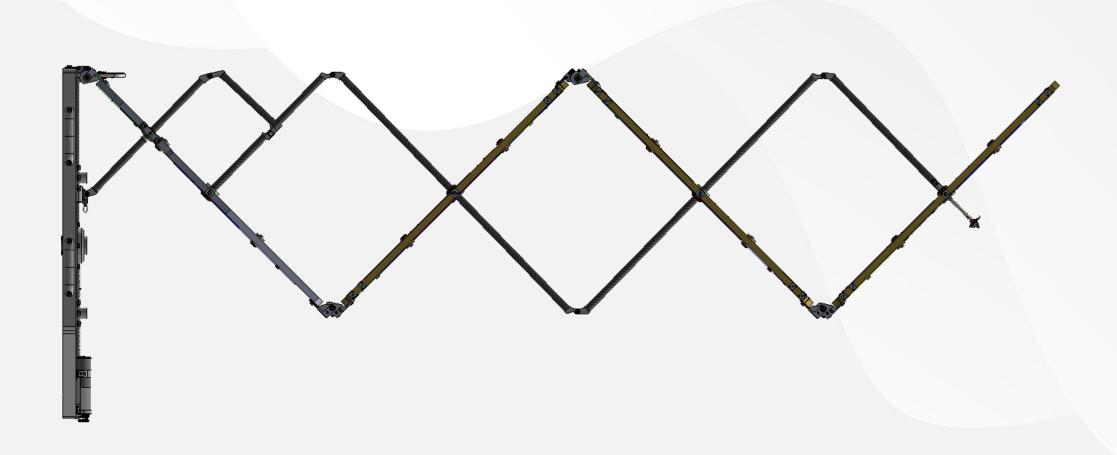
- <10 mm vending radius
- 15 tracks
- Insulation > 100MΩ at 500V
- 3,2 A per track
- ATOX protection layer: Tested at ESA ESTEC



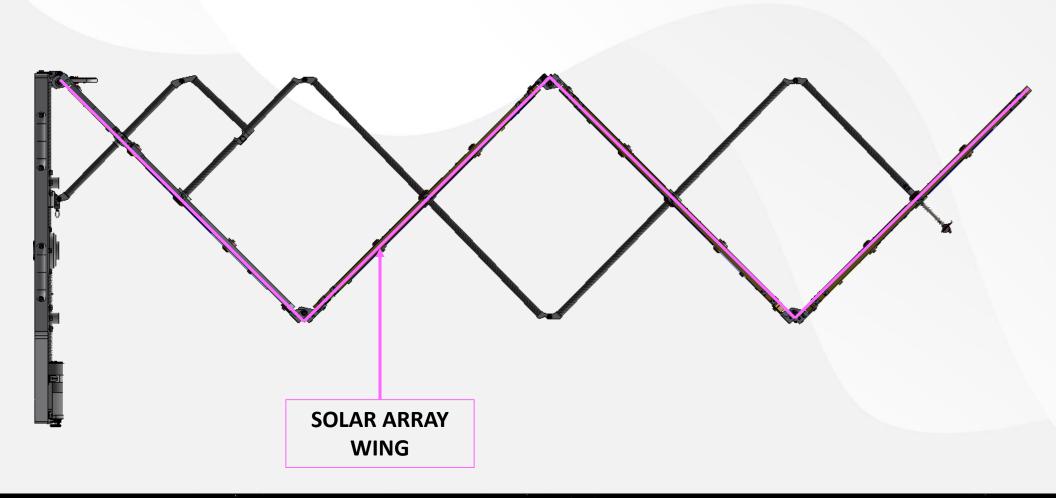


FLEXPRINT SAMPLES

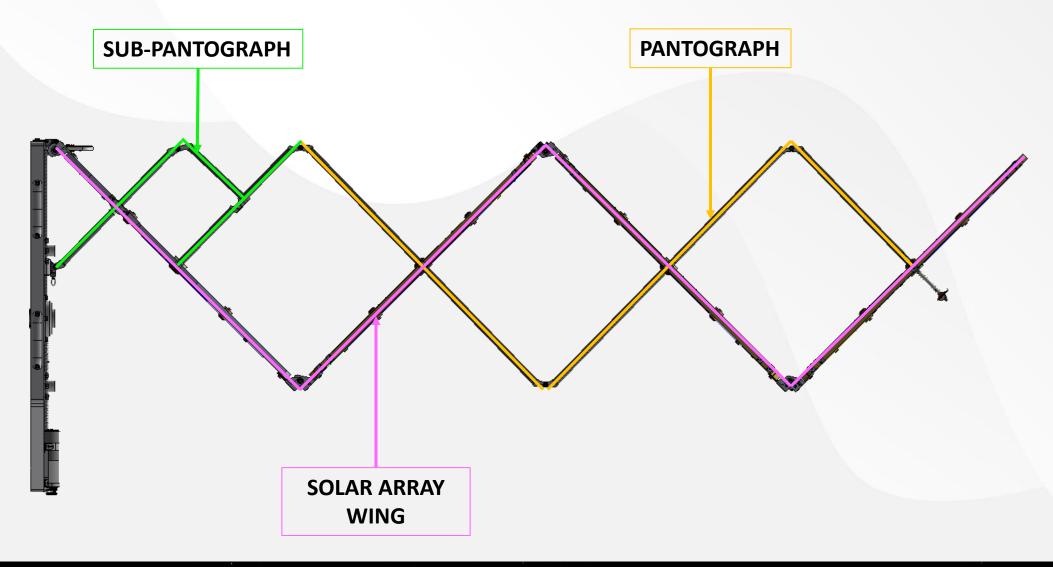






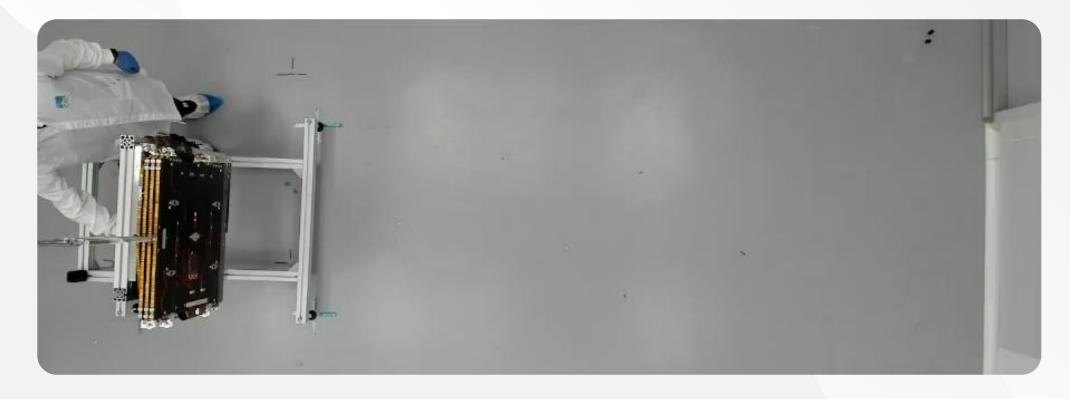






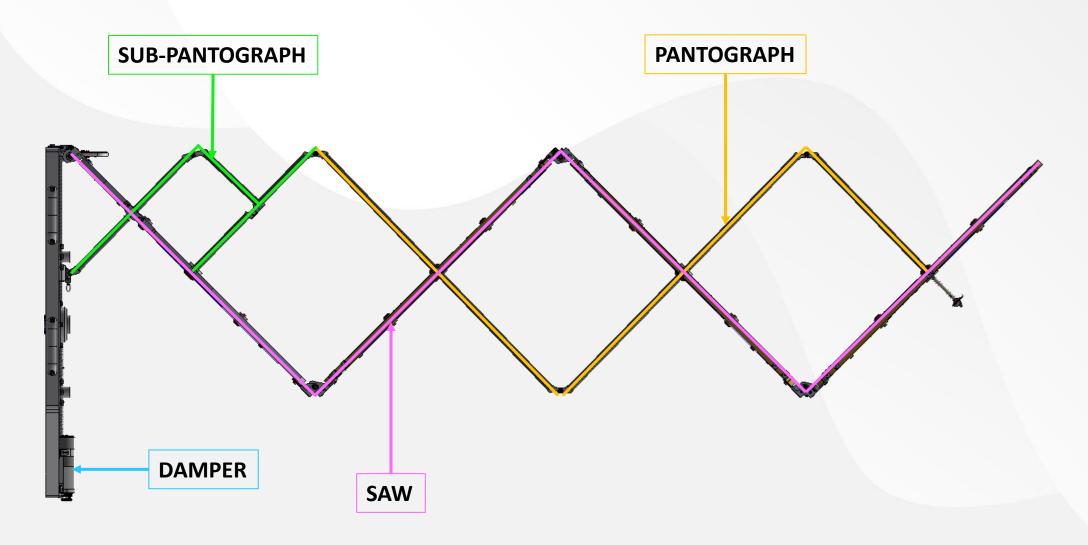


Deployment → Clocksprings driven hinges + Flexprints + Deployment Synchronization Mechanism





SAW DAMPER

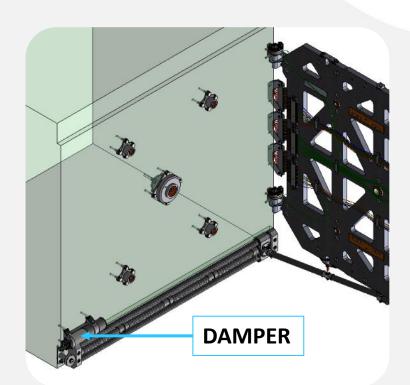




SAW DAMPER

Features:

- Eddy current damper
- Adjustable damping factor



W/O DAMPER:



WITH DAMPER:





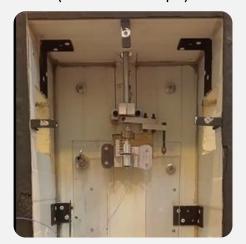
ARTICULATION THERMAL TEST

Hinge configurations:

- Hinge
- Hinge + Latching pin
- Hinge + Sensor
- Hinge + Flexprint

SENSOR

(Room Temp.)



FLEXPRINT

(Room Temp.)



LATCHING PIN

(Hot Temp.)



HINGE

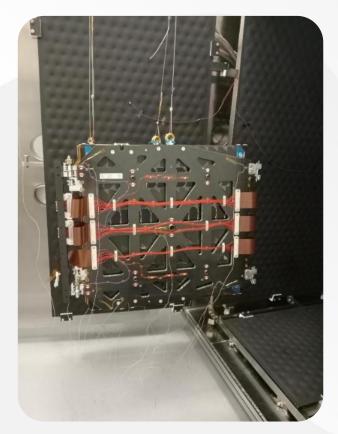
(Cold Temp.)





THERMAL VACUUM CHAMBER

- Bake out:
 - $+125^{\circ} \rightarrow 24 \text{ hours}$
- TVAC
 - $-120^{\circ}/+125^{\circ} \rightarrow 10$ cycles
 - Performed at DHV's TVAC chamber
 - Monitoring the solar cell strings electrical continuity

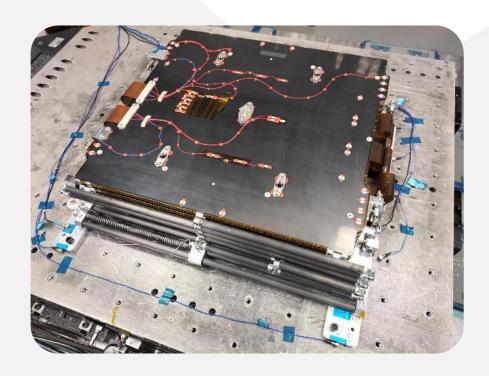


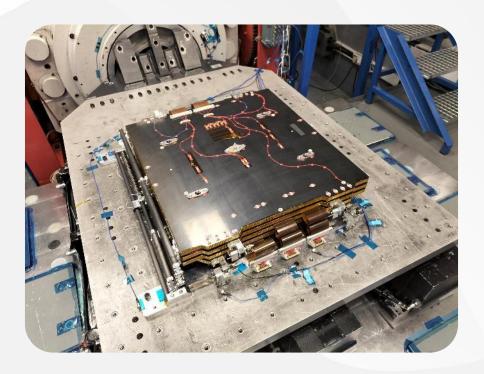




VIBRATION

- Sinusoidal vibration test
- Random vibration test→ NASA GEVS

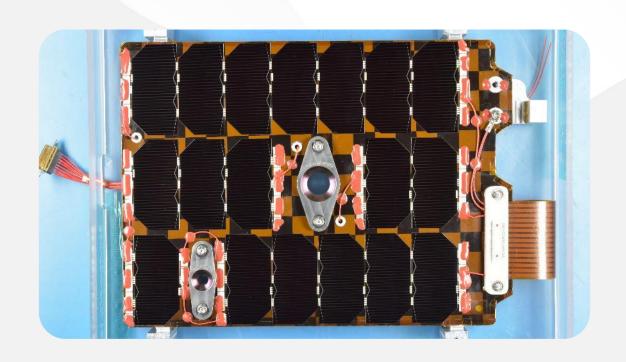


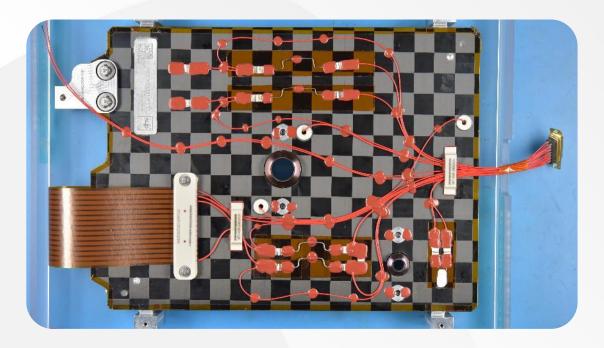




■ $-120^{\circ}/+125^{\circ} \rightarrow 1000$ cycles and beyond (LEO)

APTC Thermal Cycling







CONCLUSIONS

- New solar array wing developed for the potential SmallSat market
- Improved hinges and retention mechanisms
- Synchronized deployment using auxiliary mechanisms
- Damped deployment to avoid accelerations in the satellite's attitude
- Substrate technology developed to be modular and cost-efficient
- New electrical interconnections between panels using flexible parts
- TRL-8 to be reached before mid-2025



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