



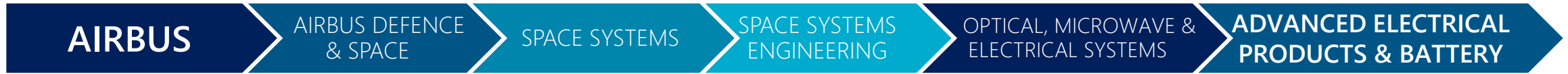
AIRBUS Space Battery Heritage Powering the Future

Space Power Workshop 2025

ALLENBACH Kévin

AIRBUS

- 1 Who we Are and What we Do
- 2 Our Products and our Heritage
- 3 Data Science for Ageing Modeling
- 4 What is next?



Europe's **number 1** Defence and Space company



More than **35 000 people** employed **worldwide**, including in the USA



One single entity active all along the Space value chain



Satellite
manufacturing



Launchers
equipment



Launchers &
launch services



Ground
segment



Satellite operations
(services)



Earth observation /
Telecoms

PIONNEERING ON BATTERY DEVELOPMENT

- **Dedicated department for Battery Development** since 2002
- **More than 20 years heritage with the Lithium-Ion (Li-Ion) technology** – 1st satellite with Li-ion battery launched in 2004 and still in-orbit!
- **New strategy** deployed in 2016:



*~ 100 employees, based in
Toulouse, **France***

From BUY...

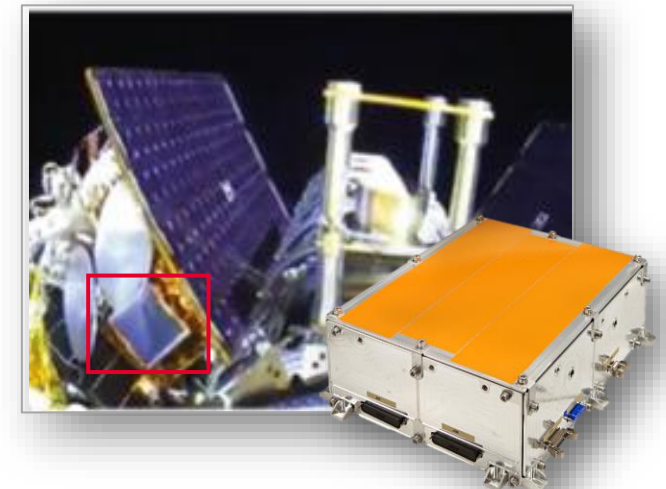
- Dedicated cells for Space
- External supply
- Airbus as “integrator”



- Costs
- Mass
- Obsolescence

To MAKE...

- COTS cells
- End-to-End battery products



3 MAIN MISSIONS



Battery Chamber of Reference for AIRBUS Group



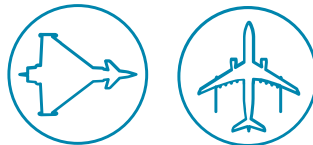
End-To-End Battery H/W products to internal/external customers **for Space, Aeronautics** and **Military** applications



Pushing **innovation** to **decrease cost & leadtime** at system level



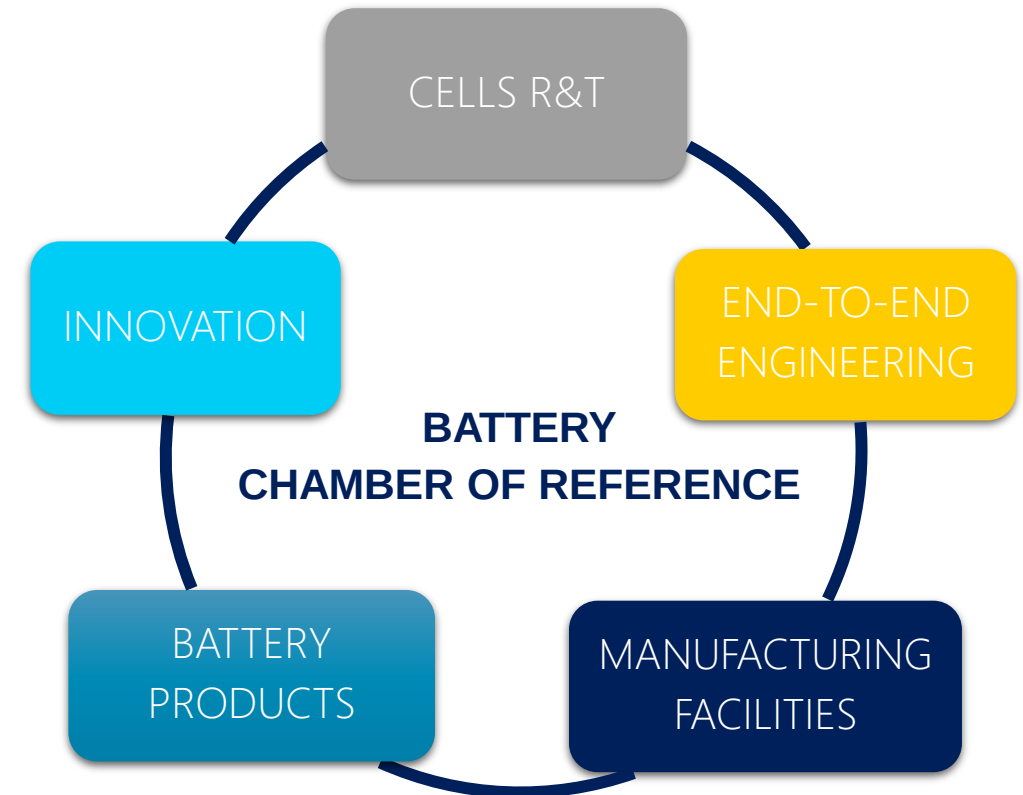
Space Systems



Military / Air Systems



Unmanned Aerial Systems

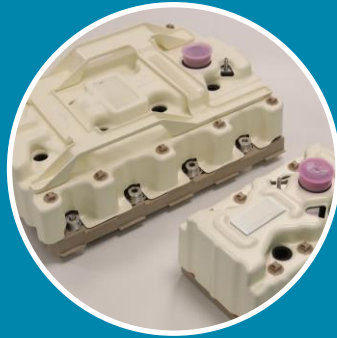


SKILLS



CELLS

Continuous electrochemical cell technology screening and testing



BATTERY DESIGN

- Mechanical and thermal design and simulations
- Electrical design and simulations



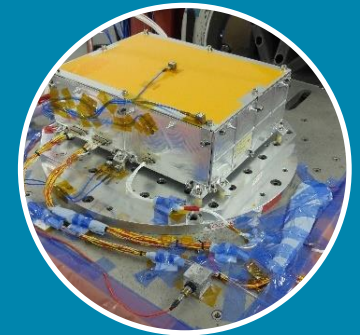
BATTERY MANAGEMENT SYSTEM

- Hardware ECSS / DO254
- Software ECSS / DO178
- Algorithms & Modelling



MANUFACTURING

- Fast prototyping
- Serial production (BAL)



TESTING

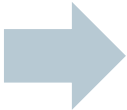
- Environmental ECSS / DO160 (thermal, mechanical)
- Electrical & Safety ECSS / DO311

END-TO-END

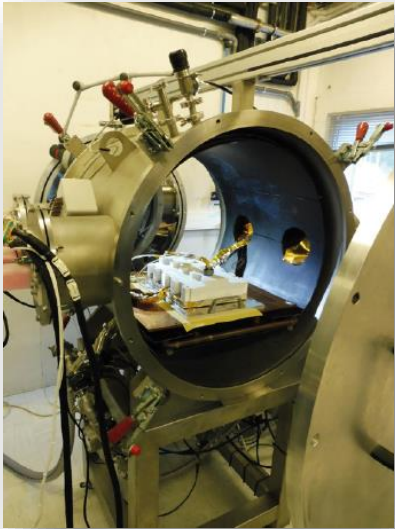
ELECTRICAL LAB



> 300 cells tests channels	20 thermal chambers -70 to +200°C
2 vacuum chambers	Abusive test facilities Fire and explosion proof equipment for cells and battery tests
1 EMC chamber ESD tests with electrostatic discharge Electromagnetic compatibility test on Space equipments	



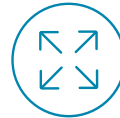
EMC/ESD
ELECTRICAL CHARACTERIZATION
SAFETY TEST
LIFE TEST



BATTERY ASSEMBLY LINE



Operated and qualified since **2019**



About **500m²** in **ISO5 cleanroom**



Lean 6-Sigma, automated process



More than **1300 modules** manufactured

4 MAIN PRODUCT LINES

BASED ON AIRBUS HERITAGE

SCAN ME TO
KNOW MORE!

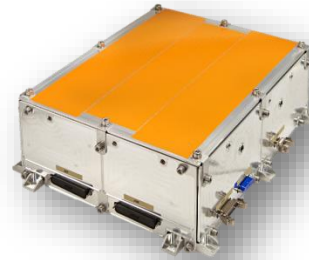


AIRBUS BATTERY KEY FEATURES

- COTS Li-ion cells qualified for Space, well established supply line (LTA)
- Industrial design & modularity
- Optimized lead-times
- Competitive REC prices while keeping a high level of reliability
- Not submitted to export licenses

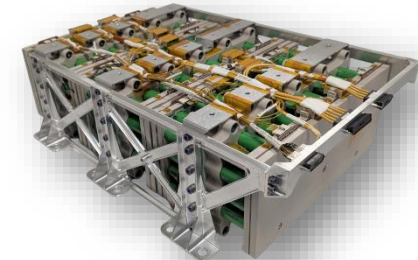
LEO CONSTELLATIONS

STELLAR-BATT



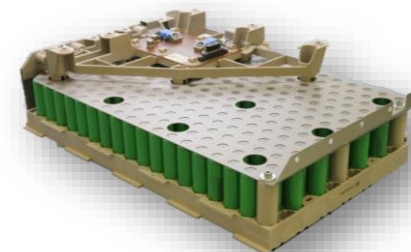
LEO OPTICAL/RADAR MISSIONS

ASTRO-BATT



GEO/MEO TELECOMMUNICATION AND NAVIGATION

COSMO-BATT



LAUNCHERS LAUNCHER-BATT



AIRBUS

28

Different battery programs

> 2000

Ordered battery modules

> 1300

Manufactured battery modules

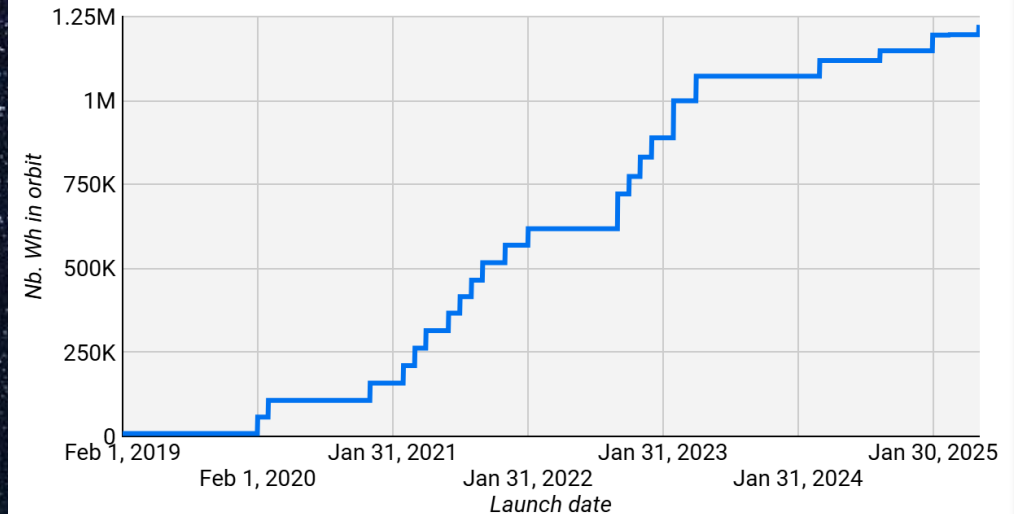
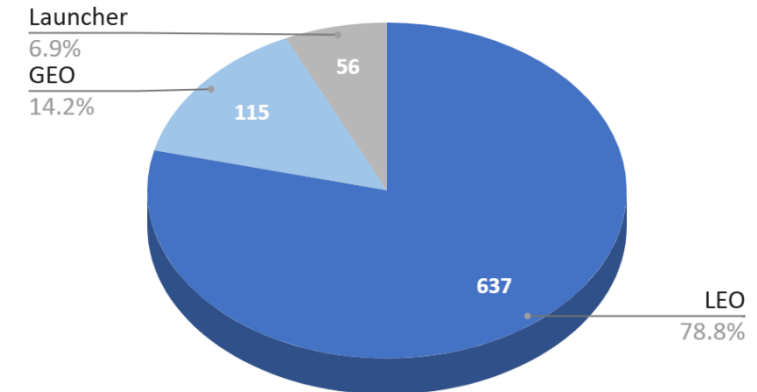
> **180k** COTS cells used, equivalent to **2,1MWh**

> **110k** COTS cells in orbit, equivalent to **1,2MWh**

> **95 Billion**

cell hours of successful
space operation with **no
reported failure**

> **800** battery modules in orbit



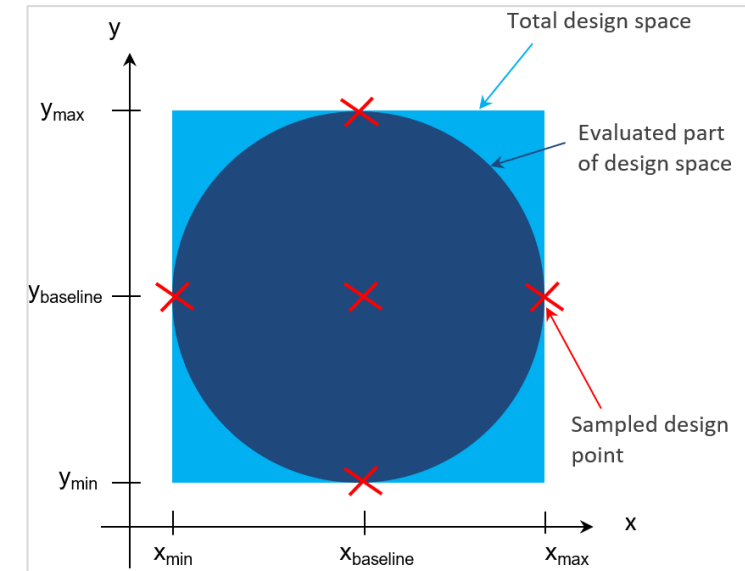
- Several parameters have an impact on **battery degradation**
- **Previous methodology** - 1 baseline test with set conditions, other tests with change of a single parameter with extrema values

Parameter
Temperature
DoD
EoCV
C-rate
D-rate

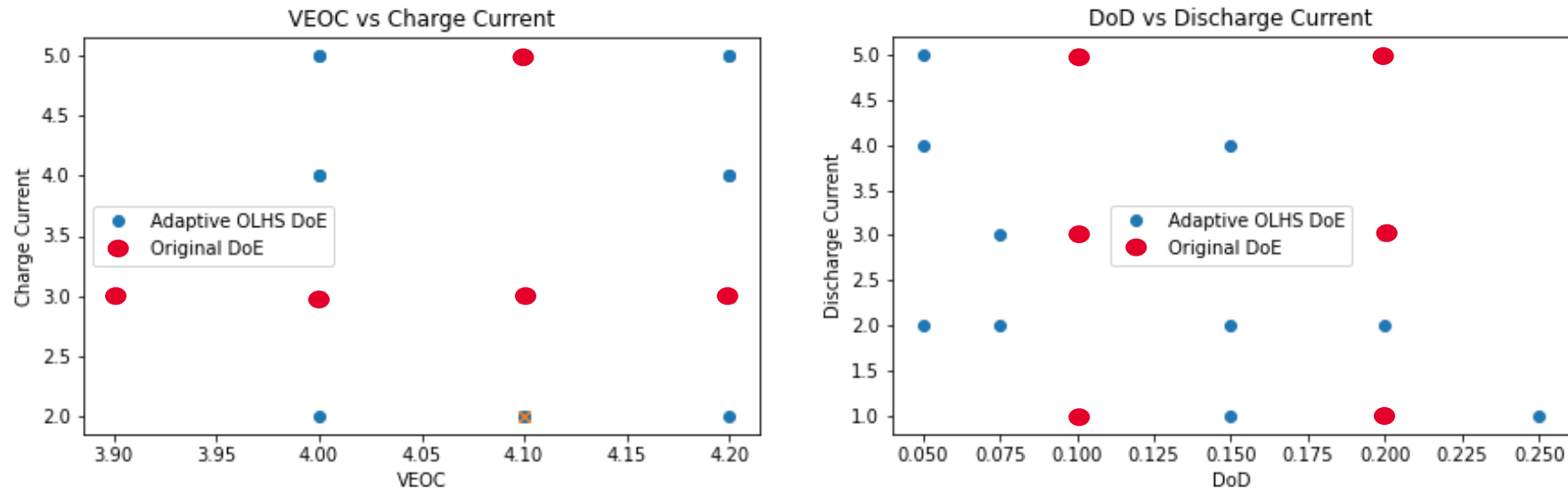
Issues: "curse of dimensionality"

- Individual impact taken into account but not the **interaction** between the parameters
- **Inability** to efficiently sample the design space in high dimensionality ($n > 3$)

➔ Necessity to deploy another approach

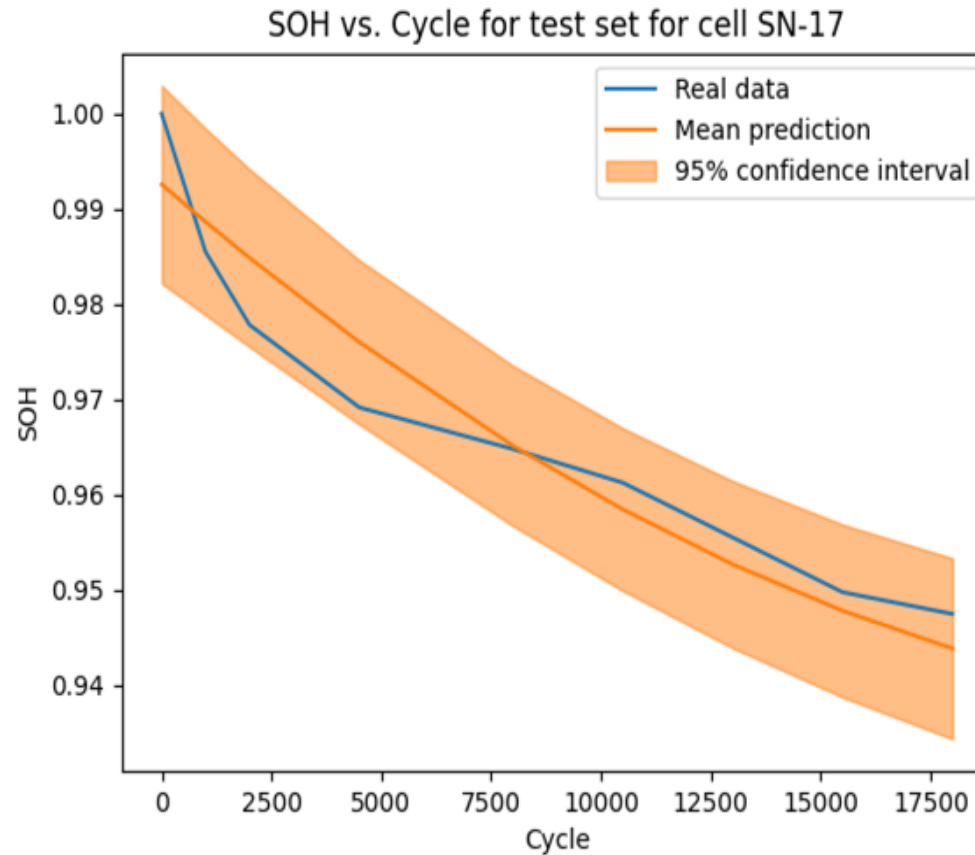


- Establishment of a **Design of Experiment**, providing an optimized sampling of the design space
- Choice of Optimized Latin Hypercube Sampling (**OLHS**) method, adaptative to take into account existing data



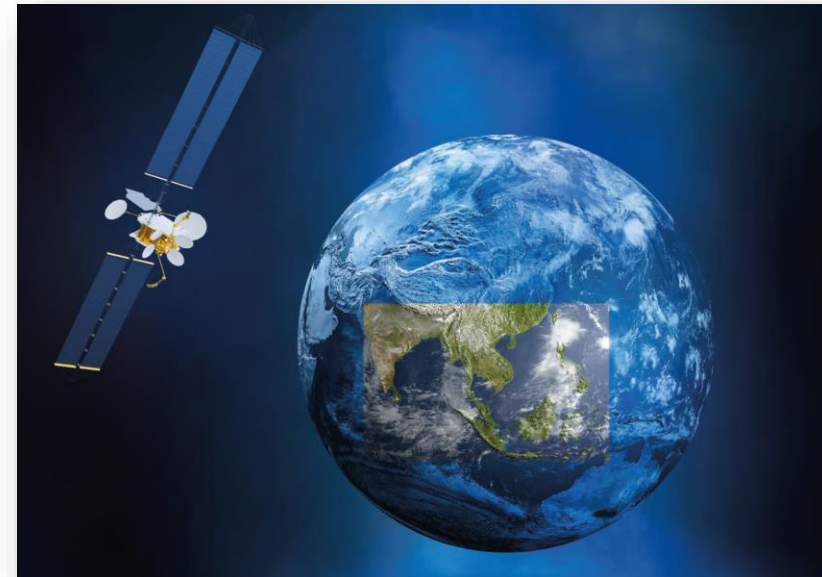
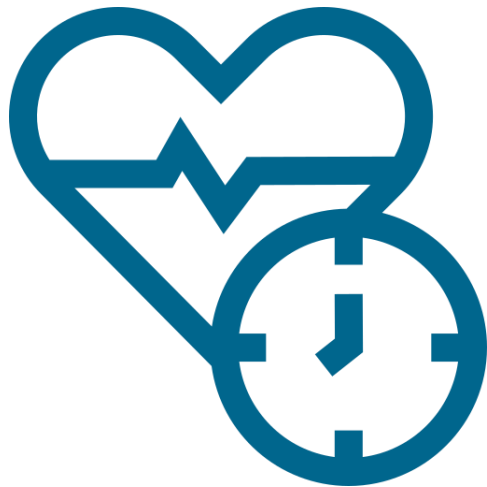
Example of test matrix deployed

- Data issued from the tests are used to develop **accurate prediction models** for battery ageing using **Machine Learning** processes.



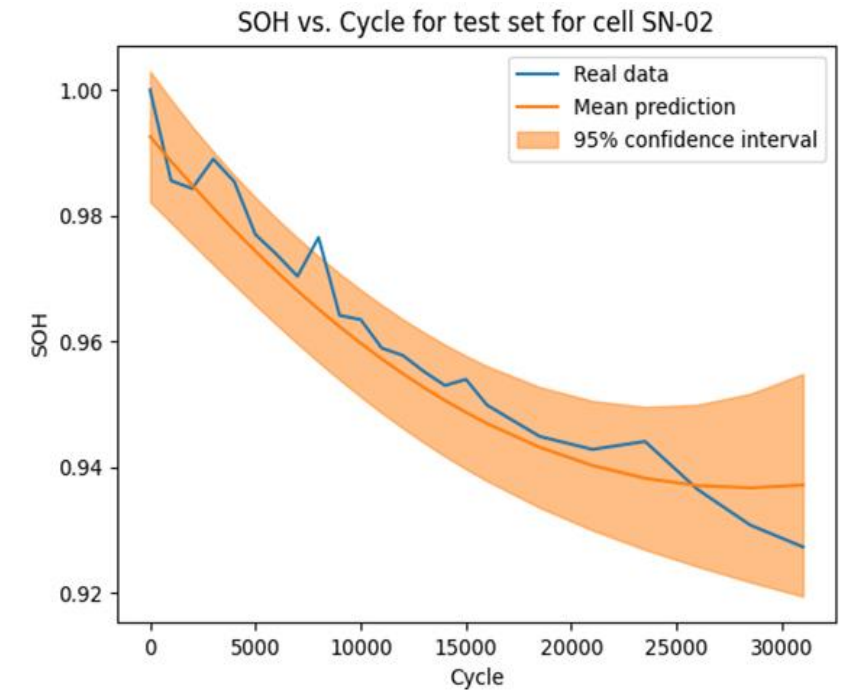
Example of prediction results

- Use case example: increase of a **battery** lifetime in orbit of **30%**
 - Lower margin as more accurate model
 - Updated ageing assessment based on customer refined mission requirements (DoD, D-rate, temperatures) and Airbus prediction model, leading to the possibility to increase the mission of **30%**



○ Limitations:

- Analysis are limited to a defined range of values
- Accuracy of the results decreases when getting closer to the boundaries of the design space
- Accuracy of the results decreases at a given number of cycles if only a limited amount of tests have reached such values
- Knee prediction only if numerous tests reach it



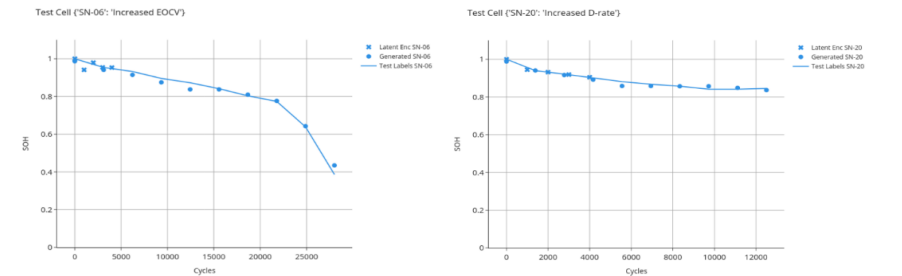
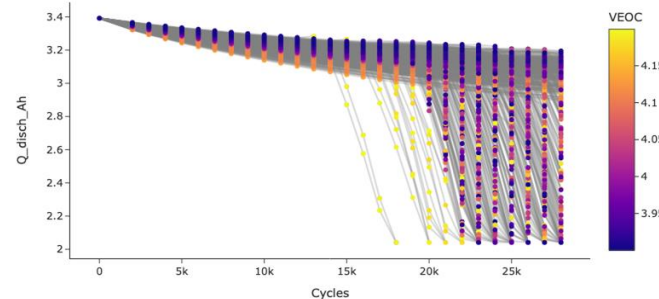
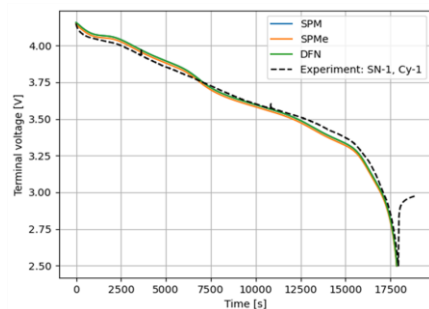
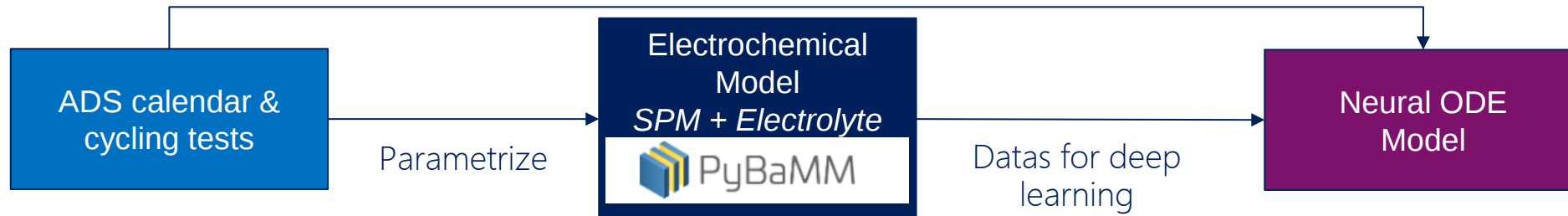
Example of prediction results

- On-going study with the University of Michigan with an **electrochemical model** coupled to **machine learning**



DATA DRIVEN → **PHYSICS INFORMED**

- Refined model with limited number of tests
- Modelisation of the knee point

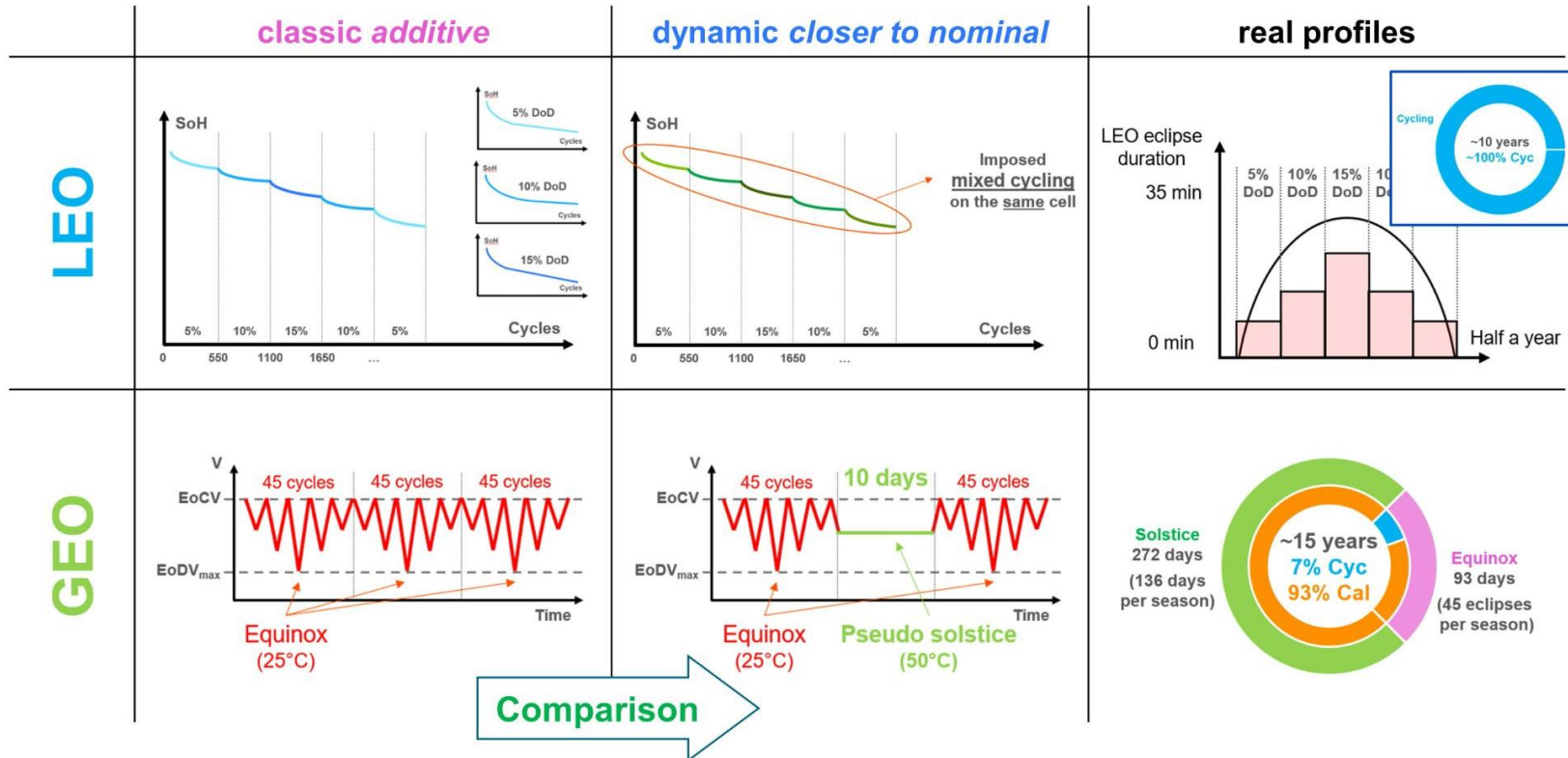


Generate Prediction

Enlarge DOE by data generation

- On-going analysis of « **Path Dependency** » impact on battery ageing

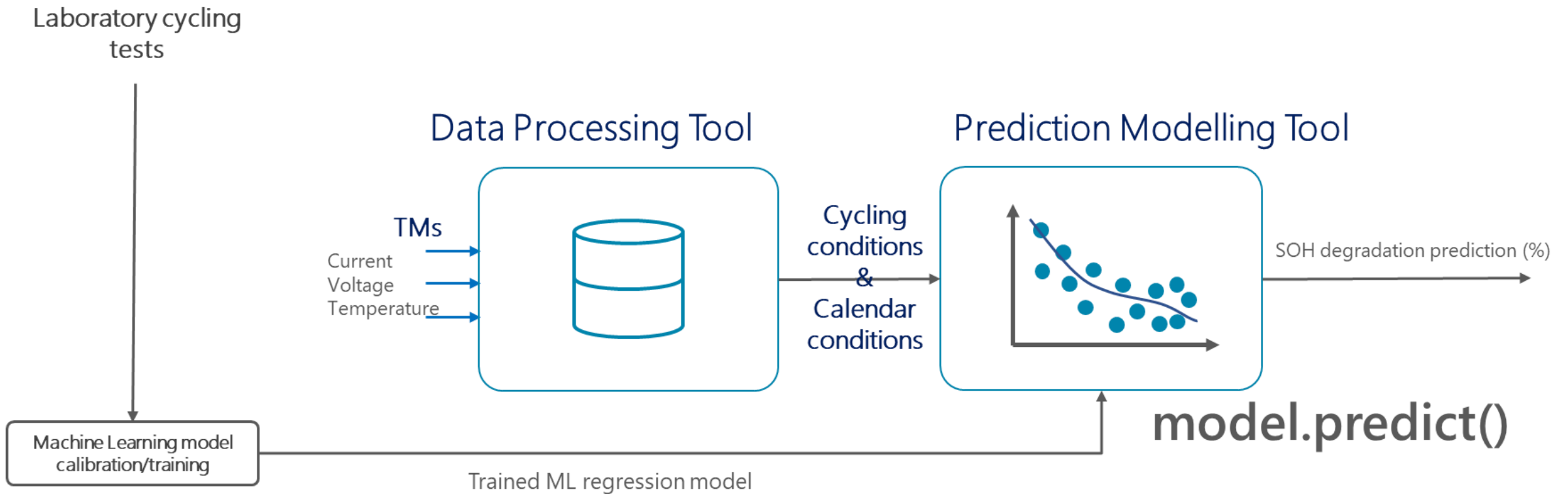
Comparison tests: Is path dependence relevant in space applications?



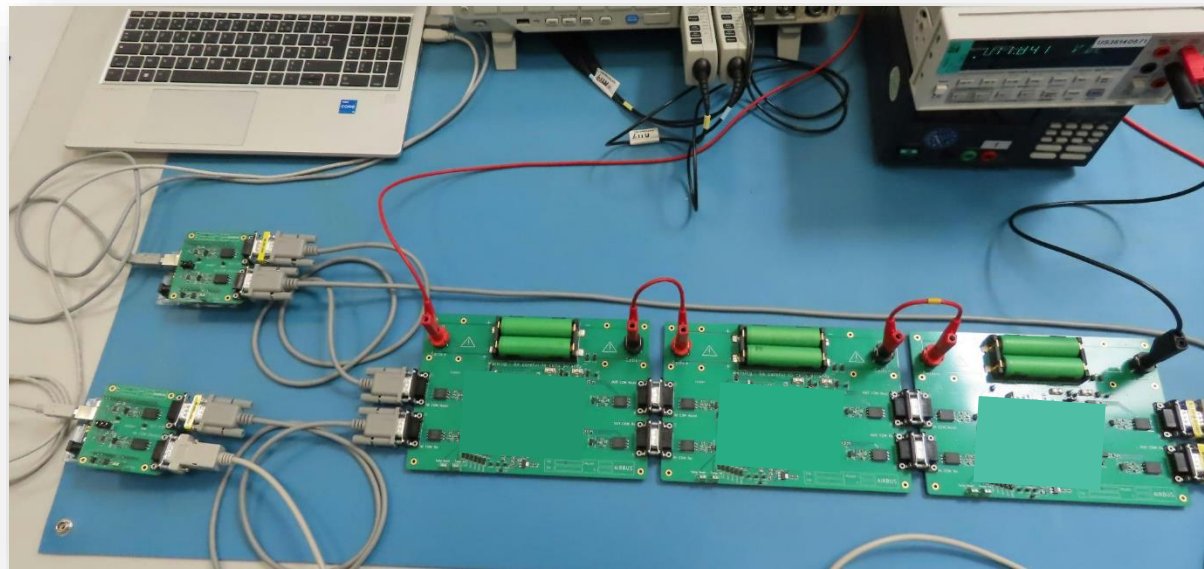
First results end 2025 –
impacts confirmed on first
results

Refined ageing model →
lowest margins

- Possibility to offer an **Ageing Prediction Tool As a Service** – enabling the customers to estimate batter degradation in EOL based on in-orbit telemetries or new use assumptions



- **Intelligent Battery** – on-going study (TRL4) considering the development of a generic electronic concept, with **embedded battery SoC and SoH estimators**; while reducing the battery interfaces to its strict minimum with digital bus CAN



Proof of Concept



ALLENBACH Kévin

Battery Business Development Manager
kevin.allenbach@airbus.com



LAURENT Julien

Battery Specialist
Julien.laurent@airbus.com



CAVAILLEZ Vincent

Head of Battery Product Team
vincent.cavaille@airbus.com



LOUP Didier

Battery Expert
didier.loup@airbus.com



SAUVAGE Damien

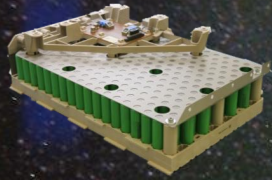
Data Scientist
damien.sauvage@airbus.com

Thank you

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SPACE BATTERY PRODUCTS | OUR PROGRAMS



EUROSTAR NEO battery
 > 160 FM modules manufactured
 > 115 modules in orbit



ONESAT battery
 > 75 FM modules manufactured



GALILEO battery
 > 75 FM modules manufactured

GEO/MEO Telecom and Nav.



OneWeb battery
 > 630 modules in orbit

LEO Constellations



FORUM battery
 Delivery beg. 2025

LEO Satellites



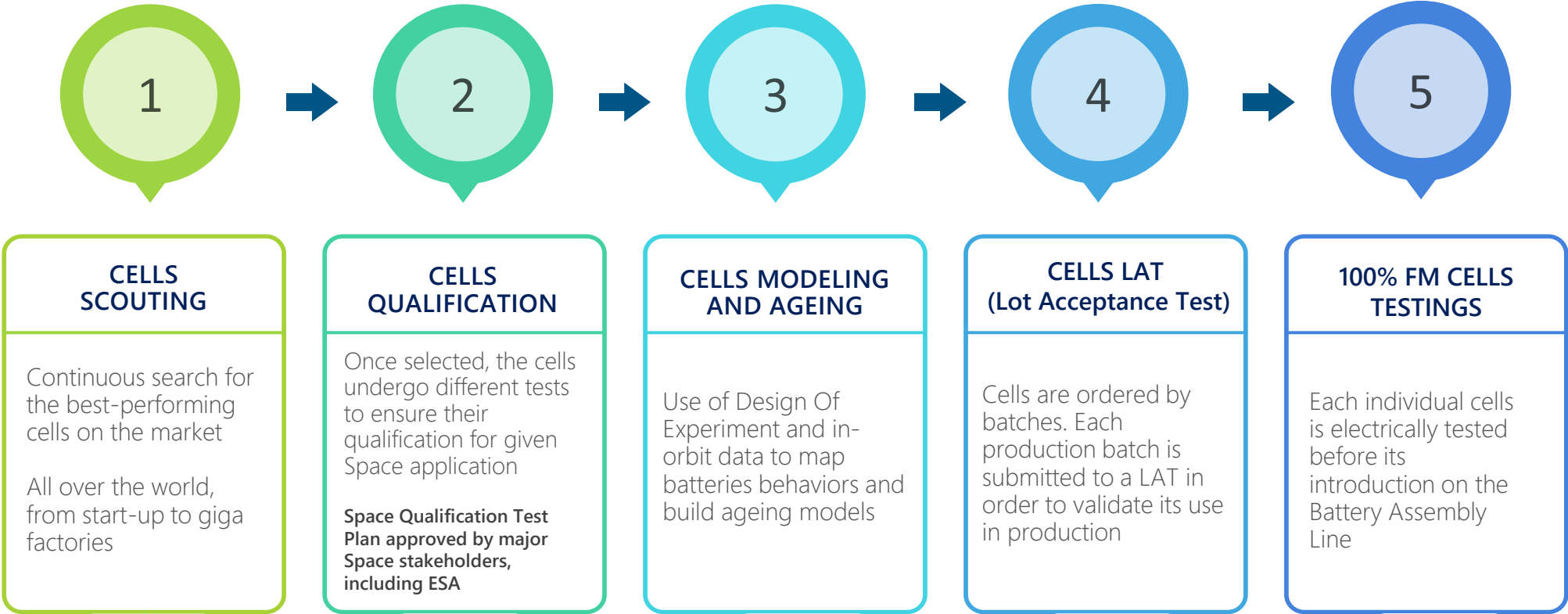
ARIANE 6 battery
 > 120 FM modules
 manufactured
 > 50 FM modules launched

Launchers

COTS CELLS STRATEGY



WE MAKE THEM SAFE !



Cycle conditions	Number of cycles reached	Equivalent years in LEO
5% DoD, C/2	110 000	19,6
10% DoD, C/3	52 000	9,3
10% DoD, C/5	43 500	7,8
20% DoD, C/5	24 500	4,4

- 65 GEO cycling seasons performed, equivalent to 32,5 years orbit

LEO ageing prediction | Correlation analysis

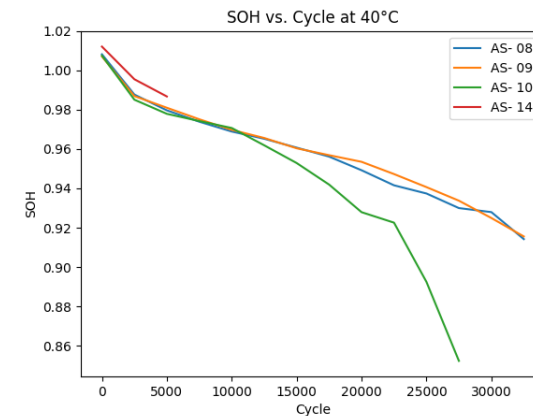
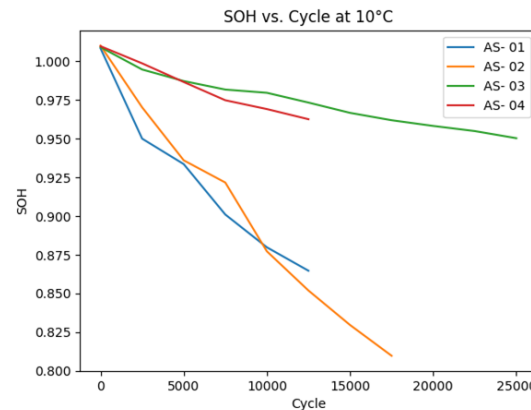
	Pearson	Pearson Partial	Spearman	Distance	Distance Partial
EOCV	-0.289680	-0.075383	-0.251482	0.092885	0.092656
DOD	-0.204948	-0.227383	-0.188036	0.057393	0.030135
D-rate	-0.512555	-0.471910	-0.396014	0.257865	0.242829
C-rate	-0.369771	-0.267153	-0.253111	0.150043	0.125070
Temperature	-0.060803	-0.009500	-0.191724	0.062747	0.050226
Cycles	-0.415048	-0.704111	-0.637155	0.307995	0.305183

- Linear correlation for cycle number => expected
- No linear correlation for temperature
- All coefficients are negative: higher values for parameters means lower SOH

- Least monotonic for DOD & Temperature

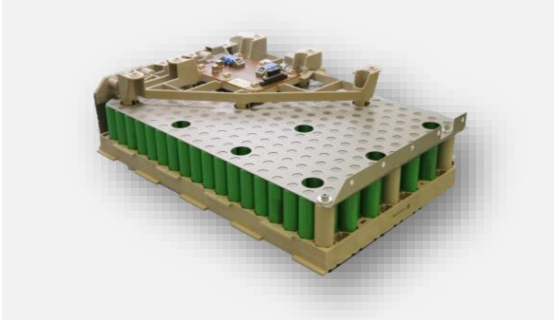
- Lower general dependency for DOD, Temperature and EOCV

- Temperature correlation difficult to highlight
- Maybe due to lack of enough data at 10 & 40°C ?



GEO/MEO TELECOM AND NAV. BATTERIES

COSMO-BATT



3 types of modules:

- **Small (-S)** – max. EoCV 4,2V, 550 to 960Wh
- **Medium (-M)** – max. EoCV 4,2V, 1100 to 2100Wh
- **Large (-L)** – max. EoCV 4,2V, 2200 to 2900Wh

Passive balancing system, normal and forced mode, voltage and temperature telemetries

LEO CONSTELLATIONS BATTERIES

STELLAR-BATT



2 types of modules:

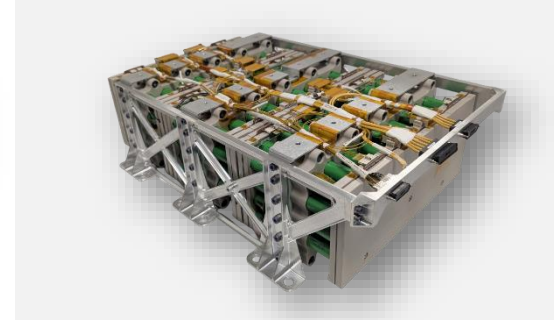
- **Small (-S)** – Max. EoCV 37,8V, 1685Wh
- **Large (-L)** – Max. EoCV 37,8V, from 900 to 3600Wh

Passive balancing, timer, power switch ON/OFF, shunt current measurement, voltage and temperature telemetries, heaters

EEE automotive COTS components

LEO OPTICAL/RADAR SATELLITES BATTERIES

ASTRO-BATT



3 voltage configurations:

- **30V** – max. EoCV 33,6V, 3600Wh (-S) or 5400Wh (-L)
- **50V** – max. EoCV 50,4V, 5400Wh
- **100V** – max. EoCV 100,8V, 5400Wh

Passive balancing system, voltage and temperature telemetries

LAUNCHERS BATTERIES

LAUNCHER-BATT



2 types of modules:

- **Small (-S)** – max. EoCV 33,6V, 260Wh
- **Large (-L)** – max. EoCV 63V, 1620Wh

Embedded fuses for safety