Space Power Workshop 2019

Torrance Marriott Redondo Beach, CA

Multi Purpose Crew Vehicle – Photo Voltaic Assembly (PVA) from development to Flight

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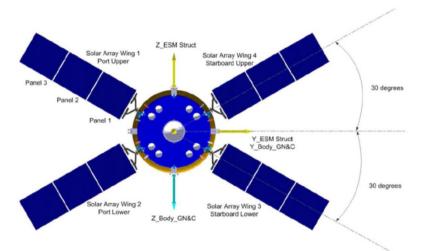






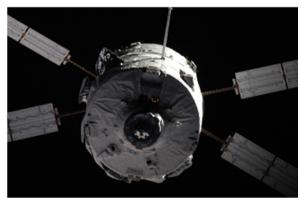


Presentation outlook



- The MPCV mission
- PVA description
- Qualification results
- Main results
- Flight manufacturing and acceptance status







The MPCV Missions (from PVA perspective)

Design Reference Mission	Specifics	Mission Time (days)	Further Requirements	Envelope Time & Radiation Environ- ment	Consolidated Mission
1. Uncrewed Lunar Flyby	LEO 185 x 1806 km Lunar Flyby Orbit (LO) 100 - 200 km	10			
2. Crewed High Lunar Orbit (HLO)	LEO 185 x 5600 km High Lunar Orbit (HLO) 1000 – 10 000 km	14	ESA.MSA.0084 "The SM shall provide active vehicle transit operations for a min. of 21 days for lunar missions"	21 days Lunar envi- ronment: Solar particle events and Galactic cosmic rays (SPE and GCR)	<u>Lunar + ISS</u>
3. Uncrewed Lunar sur- face	Low lunar orbit (LLO) 100 km	7	ESA.MSA.0002 "The SM shall support an Orion Spacecraft loitering uncrewed in LLO for at least 7 days"		
4. ISS	278 - 460 km (LEO)	216	ESA.MSA.0006 "Min. 6 days active" ESA.MSA.0007 "Min. 210 days being docked"	216 days LEO: Trapped electrons and protons, Van Allen Belts, solar particle events	



Photo Voltaic Assembly (PVA) Technology Challenges

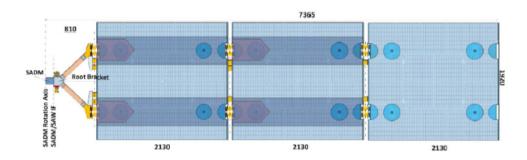
- Technological challenges
 - US standard size SCA's
 - Extreme low temperature, qualification environment for lunar mission (-205° C)
 - High bus voltage compared to heritage mission (LEO ISS)
 - Difficult definition of common "boundaries"
- Main suppliers / partners
 - Solaero (US), solar cell assemblies manufacturer
 - Airbus DS NL for the substrate
 - ESA (The Netherlands), Airbus Toulouse ITS (France) test facilities for qualification and acceptance

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MPCV Photo Voltaic Assembly (PVA)

- Leonardo responsible for design, qualification and production of PVA for the MPCV Solar Array
- Contractual customer is Airbus DS Netherlands B.V.
- Spacecraft prime is Airbus DS Bremen
- Solar cell assemblies are CFI from Airbus LMC (Solaero)
- NASA ESA joint program
- PVA Key facts
 - 14904 solar cells installed on a total area of 49 m² area
 - 12 panels 2,1 m x 1,9 m each, ~ 70 kg mass x wing (including substrate and mechanisms)
 - More than11 kW EOL SS developed power @ 125 V bus voltage



Courtesy of ADSNL

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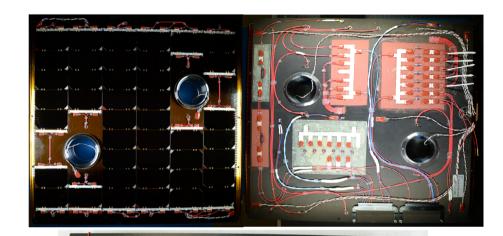


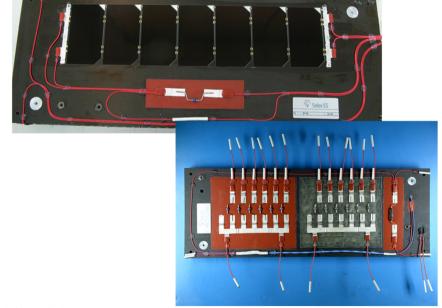
Main results

- Main achievements
 - Qualification for the worst case lunar environment completed
 - QM model successfully delivered
 - ESM1 PVA successfully delivered and SA under S/C integration
 - ESM2 PVA under manufacturing
- Next events
 - ESM2 acceptance test campaign to be completed within Q4 2019
 - Likely change of the thermal vacuum temperature acceptance profile
 - Delta qualification for covering future LEO missions



DVT models qualification





- A number of DVT coupon have been manufactured for qualifying baseline and backup technologies
- Issues discovered at the begining of the qualification campaign were covered by delta actitivies
- Extensive TV cycles
 - 10 cycles @ -214° C/+150° C
 - 350 cycles @ -177° C/+150° C
 - 13 cycles @ -214° C/+150° C
- Simulation of long term storage conditions
 - 30 days @ 60 ° C and 90% RH



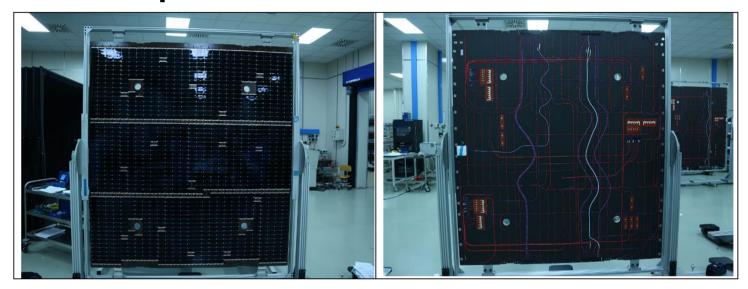
QM panels



- Partly integrated by Leonardo (PVA only)
- Submitted to a full mechanical environmental qualification
- Integrated onto the MPCV Qualification Model S/C
- Not 100% representative because of the change of the substrate manufacturing technology (Airbus proprietary)



ESM1 and 2 panels

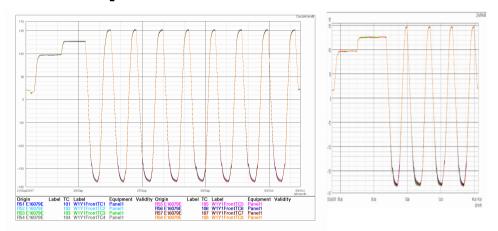


- Three sections per panel
- ESD safe design (limitation of worst case maximum differential voltage)
- 12 panels delivered in batch of 3 (ESM 1)
- 6 panels under manufacturing (ESM 2)

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ESM1 panels





- 12 panels delivered in batch of 3
- (P)FM and FM acceptance sequence
- TV acceptance test successfully completed and final inspection and reparing performed on Customer site
- Extensive use of ESA portable flasher system for power performance comparison pre and post TV test



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